Principali attività di ricerca in corso e possibili sviluppi

Daniele Bianchi *Dipartimento di Ingegneria Meccanica e Aerospaziale*

Incontro promozione attività di ricerca DIMA Roma, 6 febbraio 2017





Main activities

- Solid rockets:
 - Nozzle flows coupled with thermochemical ablation modeling
 - Nozzle transient heating and shape evolution due to ablation
- Hybrid rockets:
 - Modeling of motor internal ballistics and performance
 - Modeling of gas-surface interaction
 - fuel pyrolysis
 - nozzle ablation
 - liquefying fuels
 - Reentry vehicles and hypersonics:
 - Ablative thermal protection systems (TPS) modeling and simulations
 - Models for ablation, pyrolysis and catalysis over TPS



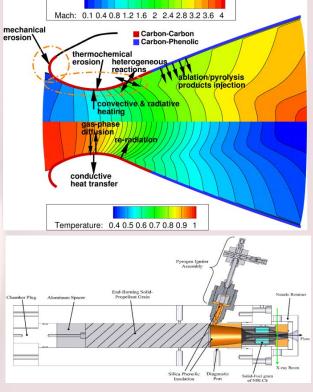
Nozzle thermochemical ablation in solid rocket motors

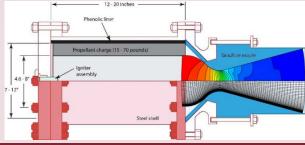
- DIMA personnel (all activities below): Bianchi, Nasuti, Onofri
- Collaborations: Vigor Yang (Georgia Tech) Kenneth Kuo (Penn State)

B. Favini E.

Cavallini (DIMA)

- Background (since ~ 2004):
 - Coupled analysis of flow and surface ablation in carbon-carbon rocket nozzles
 - Extension to finite-rate ablation modeling, shape change, and pyrolyzing materials (e.g. carbonphenolic)
 - Effect of radiation and roughness (V. Yang), high pressure (K. Kuo), and nozzle shape
- Recent activities (ESA-ESRIN 2014/2015):
 - Study of nozzle erosion behavior of the carboncarbon nozzles for the three solid stages of Vega launcher (P80, Zefiro 23 and Zefiro 9)

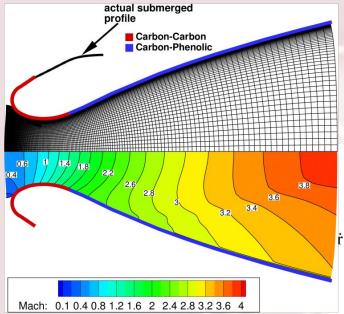




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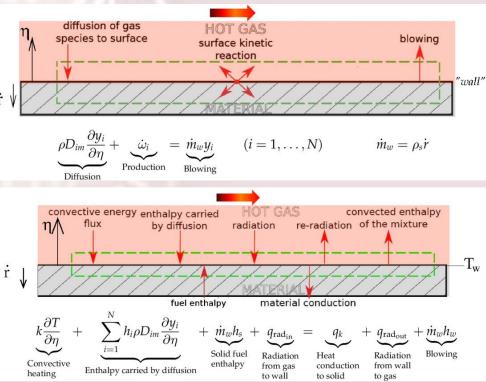


Nozzle thermochemical ablation in solid rocket motors



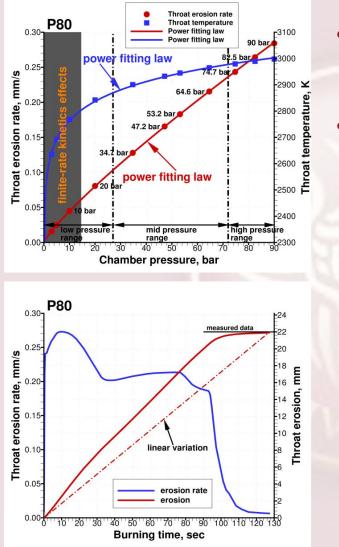
- Surface ablation modeled via finite-rate heterogeneous reaction mechanism
- Thermal conduction and radiation are accounted for

- Thermochemical erosion calculation as part of flow-field solution
- Detailed surface mass and energy balance at the ablating surface

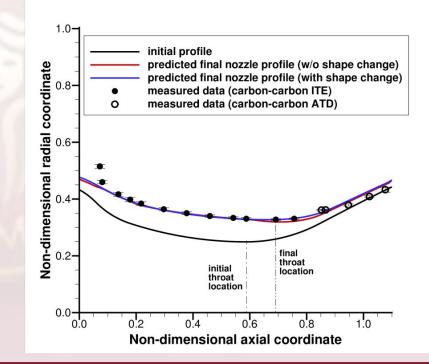




Nozzle thermochemical ablation in solid rocket motors



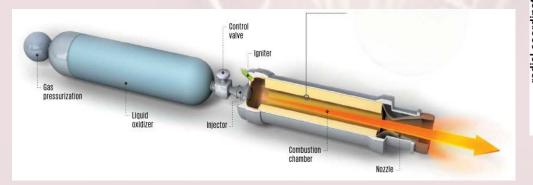
- Thermochemical ablation characterization in terms of nozzle throat erosion dependency upon motor chamber pressure
- Reconstruction of nozzle throat history and time evolution of nozzle profile

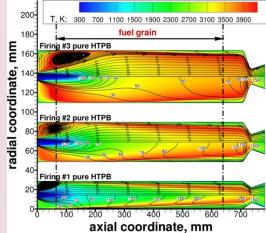


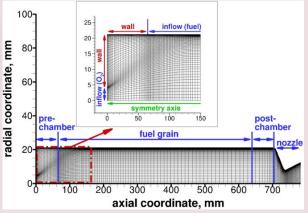


Numerical Simulations of Hybrid Rockets

- DIMA personnel (all activities below): Bianchi, Nasuti
- Collaborations: R. Savino, C. Carmicino (UniNa Federico II), L. Galfetti (PoliMi), D. Pastrone (PoliTo)
- Background (since ~ 2009):
 - Nozzle ablation modeling in hybrid rocket environment
 - Gas-surface interaction modeling for fuel regression
 - Extension of modeling approach to liquefying fuels
- Recent activities (PRIN 2009):
 - Development and integration of the Italian scientific expertise for the progress of hybrid rocket engines

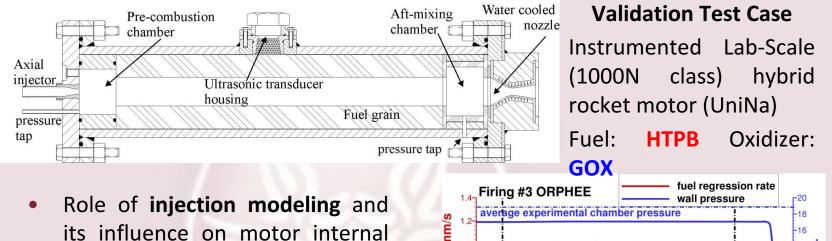






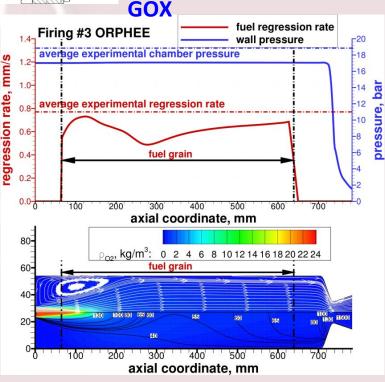


Numerical Simulations of Hybrid Rockets

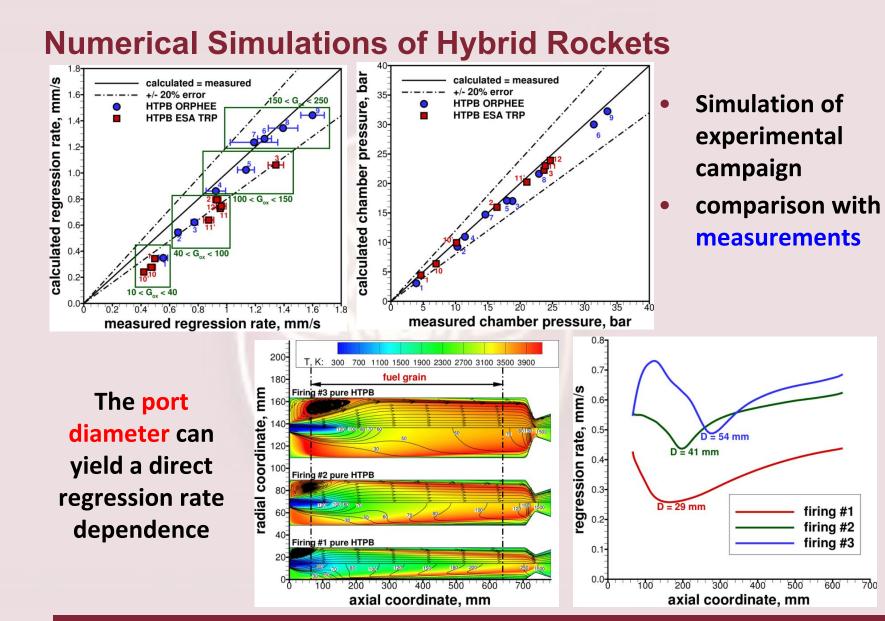


ballistics

- "full inlet" vs "injector" configuration
- Evolution of recirculation region inside the fuel port
- Effect of recirculation region on fuel regression rate and its influence on the motor performance

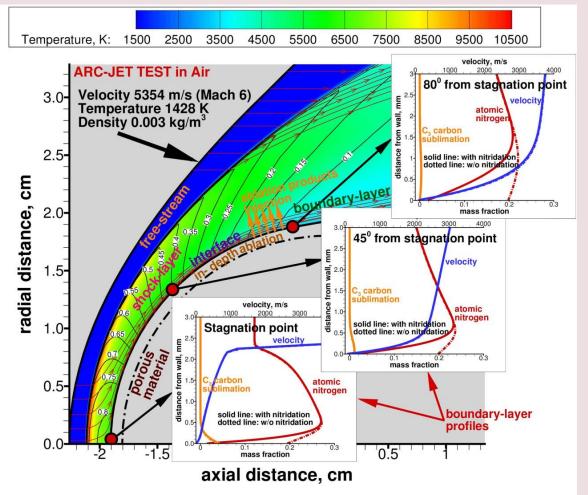






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Ablation/Catalysis modelling for high speed flows



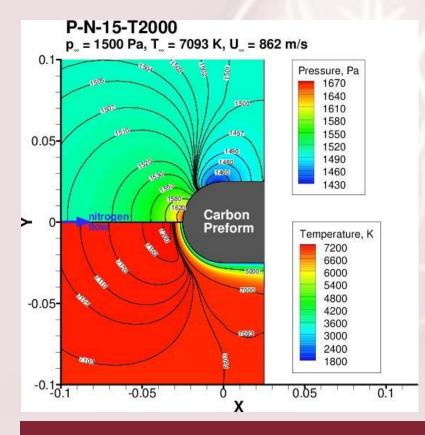
The availability of CFD tools with integrated **Gas Surface Interaction (GSI) capabilities** is fundamental in order to perform testing and evaluation of ablation and catalysis models:

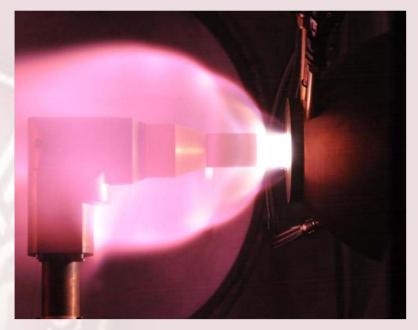
the gas interacts with the surface in such a way to alter the gasphase solution through chemical changes and/or surface blowing



Ablation/Catalysis modelling for high speed flows

Catalytic properties of ablators ESA ITT AO/1-7664/13/NL/RA UniRoma (IT), VKI (BE), CFSE (CH), CNRS-LCTS (FR), Astrium ST (FR)





Goals:

- Material **ablation** and **catalycity** phenomenological and physical models implementation, testing and evaluation
- models refinement, verification, and validation with PWT data

